



# Status report and lessons learned from the Univ. of Arizona NMSD

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# Lightweight Mirror Requirements

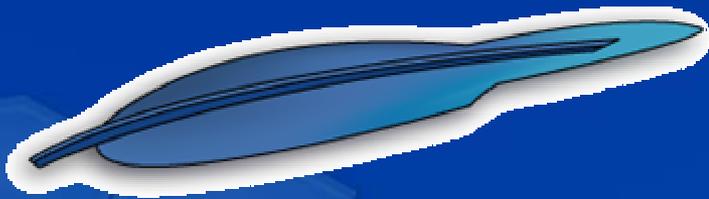
The next generation of space mirrors need to meet the following criteria:



Accurate



Easily stowed



Feather-light

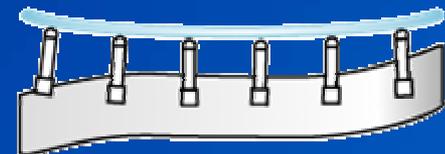
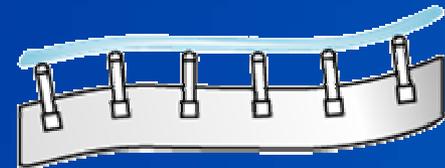
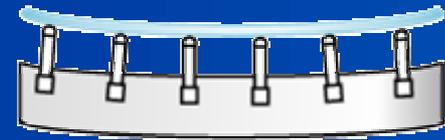


Robust

# Arizona's Active Mirrors

## Components

1. Glass facesheet
2. Position actuators
3. Lightweight reaction structure

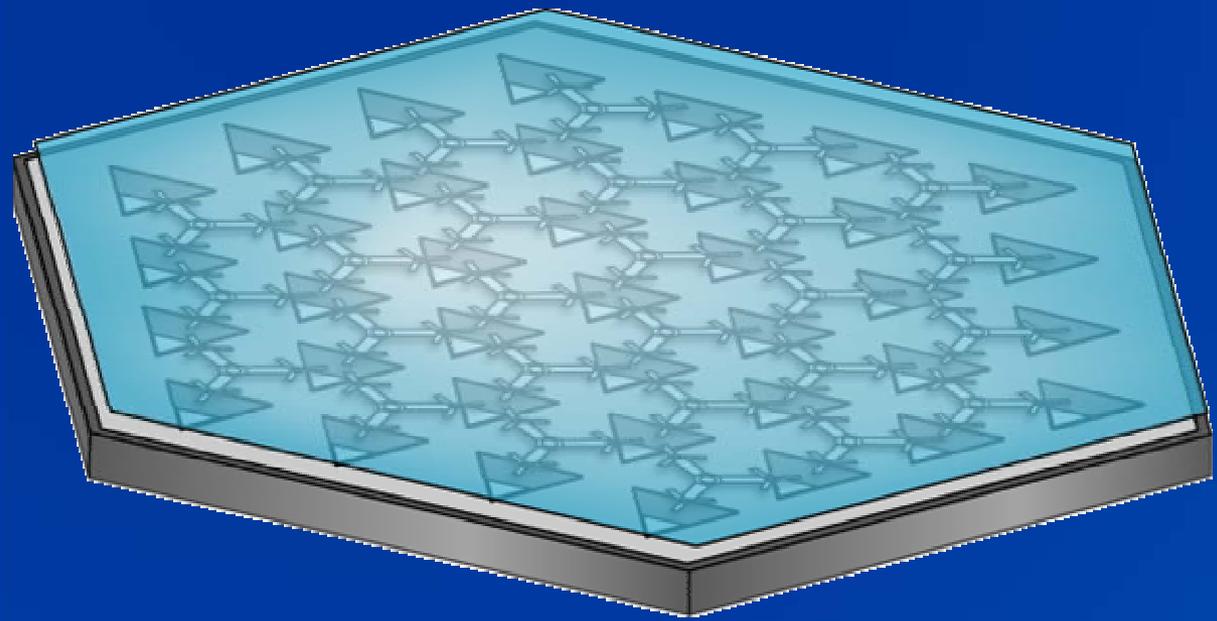


## Operation

1. Ideal Shape
2. Structure deforms, membrane follows
3. Actuators correct figure

# UA 2-m Prototype

Originally designed as NGST technology



- 2 meters in diameter
- 2 mm thick facesheet
- 166 actuators
- 35K operation
- 13 kg/m<sup>2</sup>

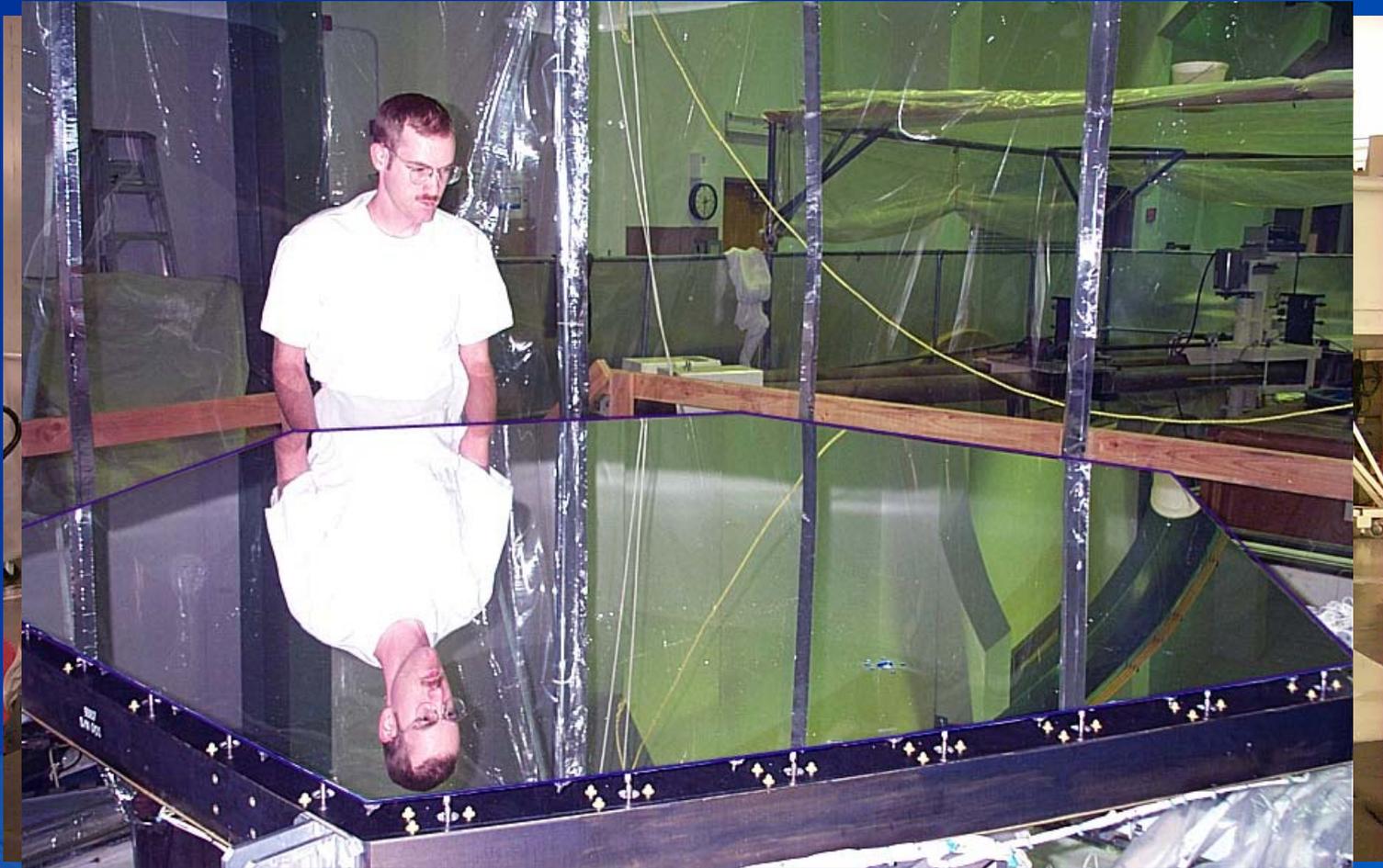
**86 pounds!**



UA LIGHTWEIGHT  
MIRROR TECHNOLOGY

# UA 2-m Prototype

Deblocked Assembled Mirror (2 mm thick)



# Metrology Schedule

The mirror does not assume the proper figure when initially assembled

Phase 1

Phase 2

Phase 3

Hartmann  
Test

IR  
Interferometry

Visible  
Interferometr  
y



INCREASED MEASUREMENT ACCURACY



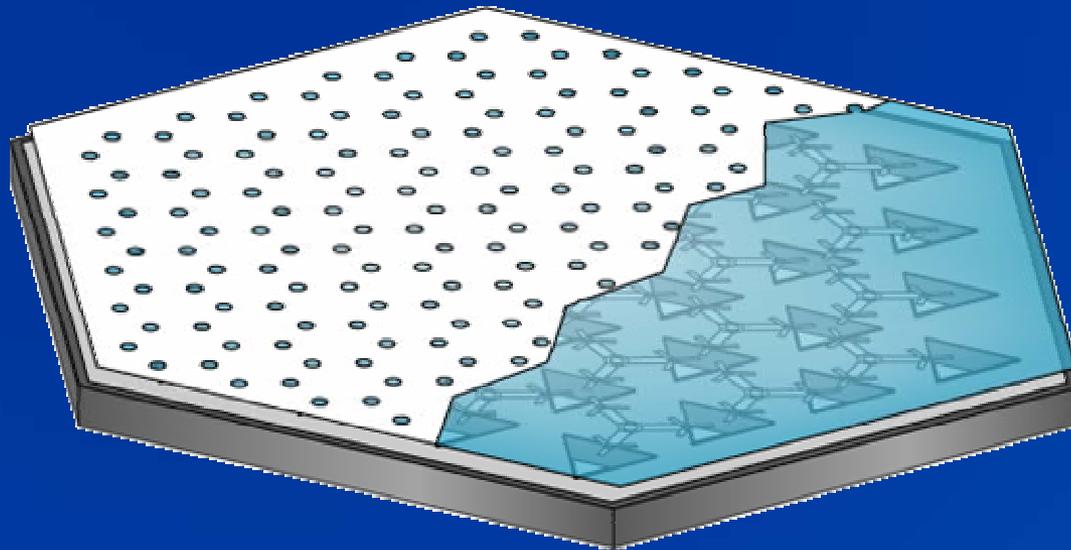
INCREASED DYNAMIC RANGE



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# The Hartmann Test

## Test Setup

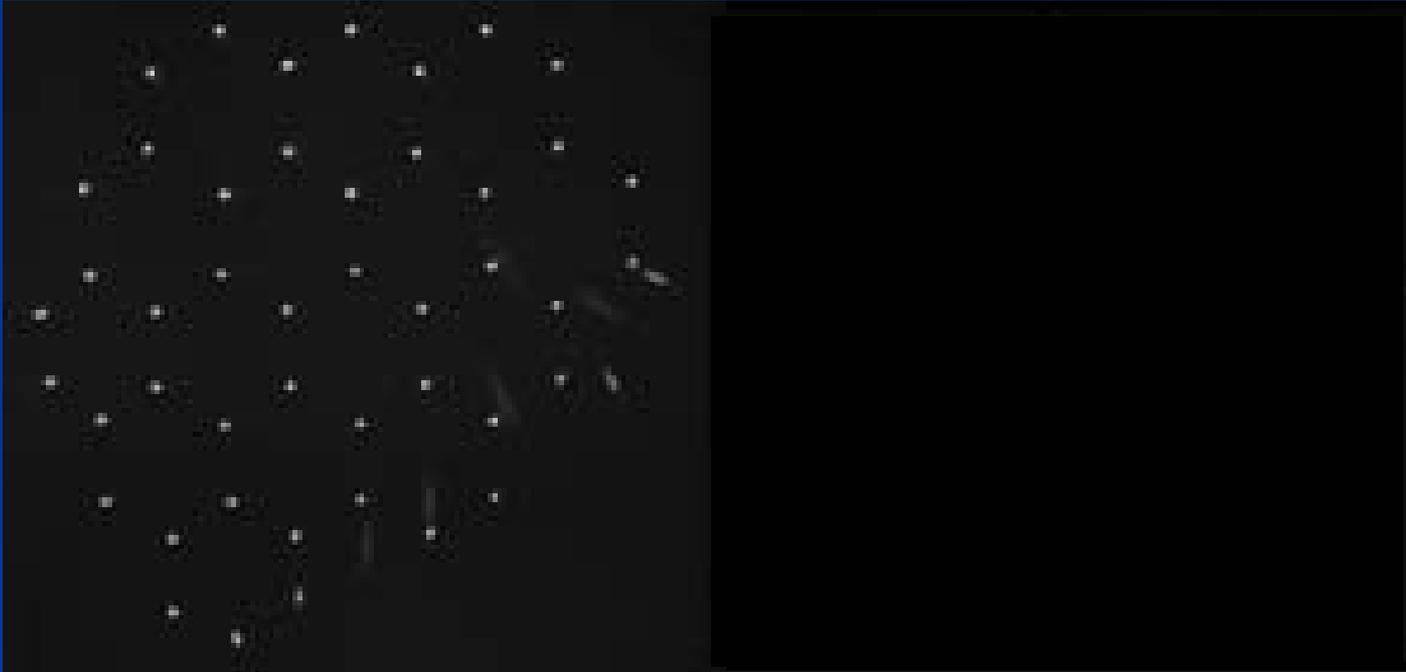


Paper mask has  
216 holes

Hartmann test has  
a large dynamic  
range

# The Hartmann Test

## Step 1: Qualitative Testing Results



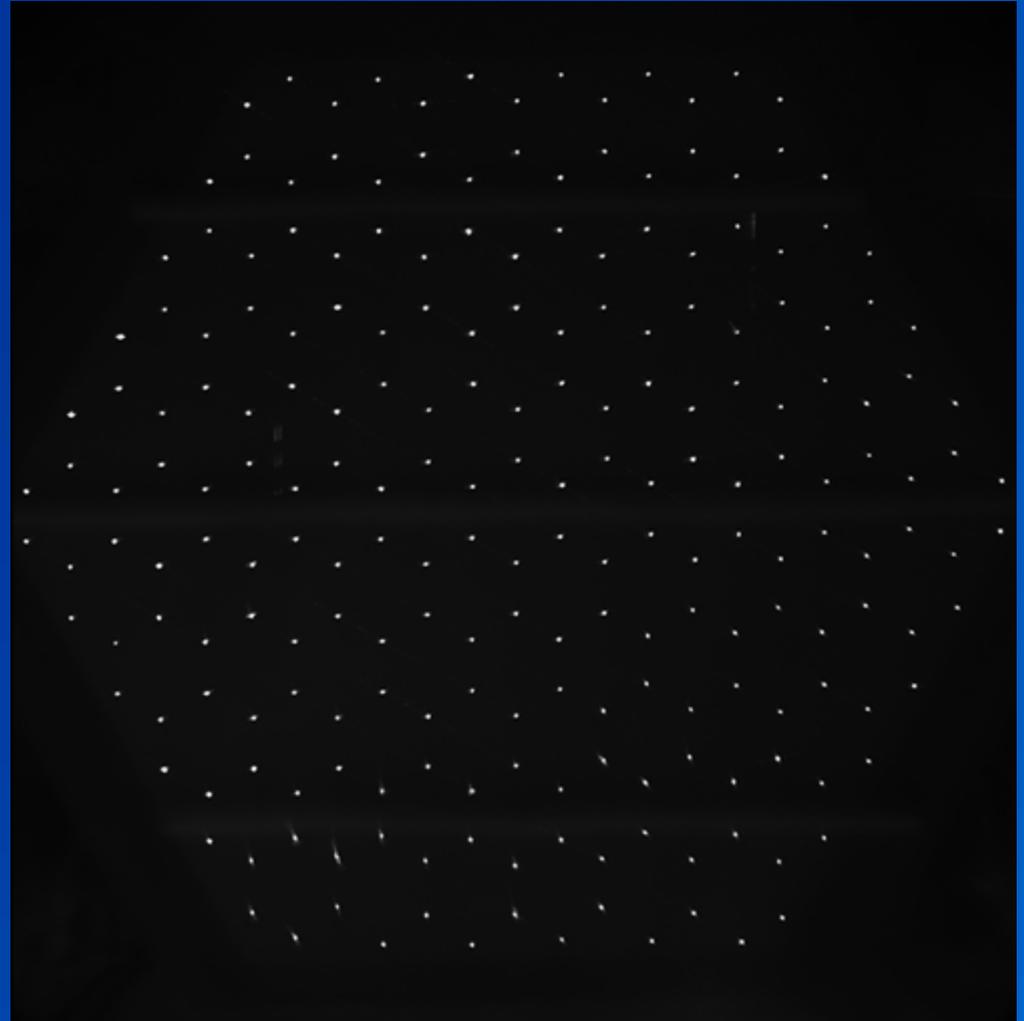
Adjust by hand until a set of six spots  
appear around every actuator

# The Hartmann Test

## Step 2: Quantitative Testing

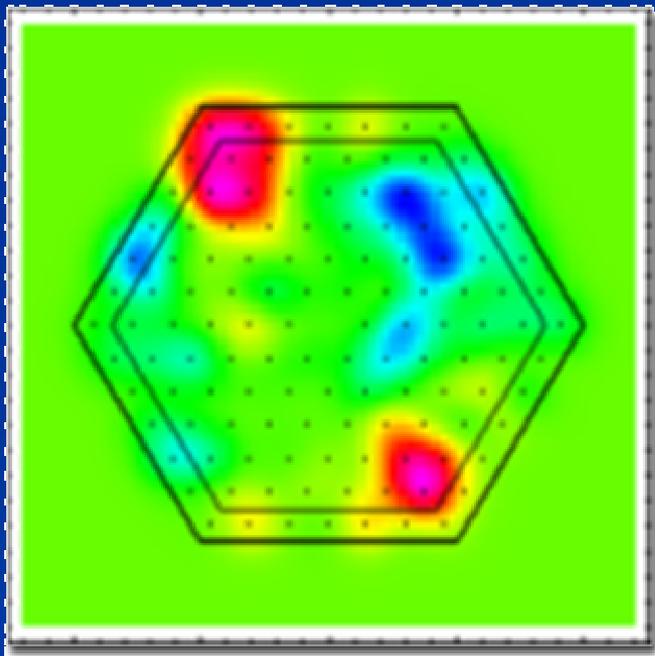
Hexagons that get  
**larger** contain  
actuators that are too  
**high**

Hexagons that get  
**smaller** contain  
actuators that are too  
**low**

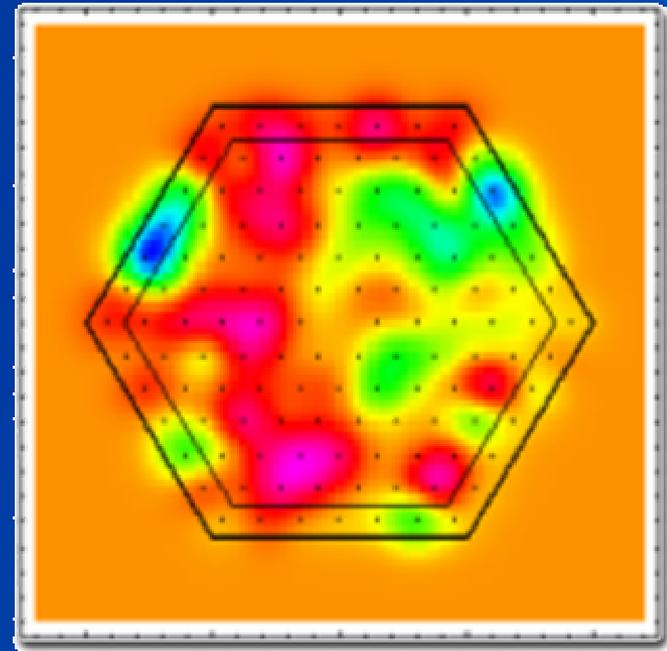


# The Hartmann Test

Step 2: Quantitative Testing – Initial & Final Results



125  $\mu\text{m}$  PV  
25  $\mu\text{m}$  RMS



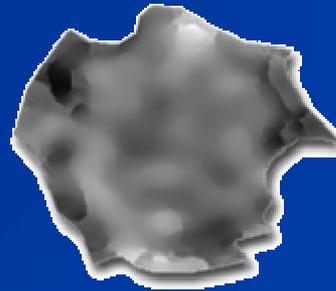
23.1  $\mu\text{m}$  PV  
3.7  $\mu\text{m}$  RMS

# IR Interferometry

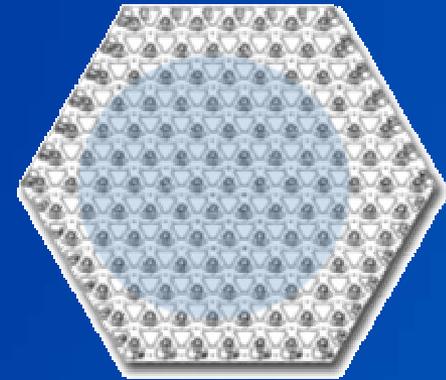
## Final IR Measurement Results



IR Interferogram  
(10.6 microns)



Surface Map



Area under test

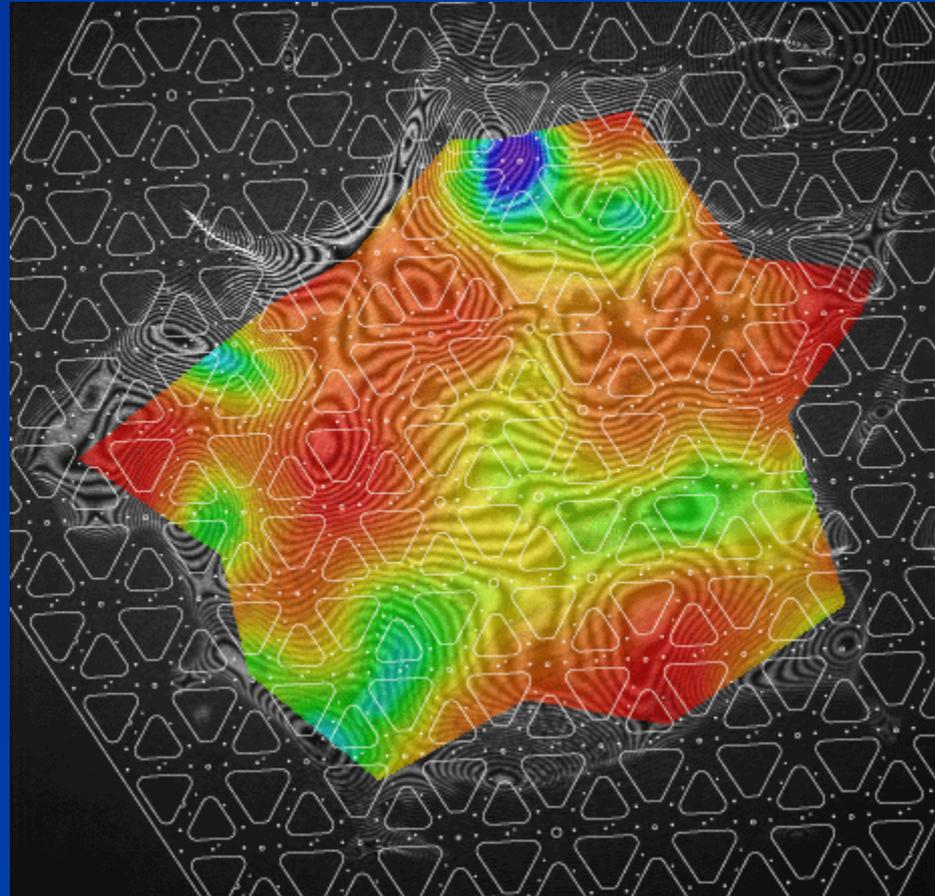
1.4 wvs PV / 0.17 wvs RMS (10.6 um)

# Visible Interferometry

Initial visible (HeNe) measurement

14 wvs P-V  
2.0 wvs RMS  
(HeNe 633 nm)

8.9  $\mu\text{m}$  P-V  
1.266  $\mu\text{m}$  RMS



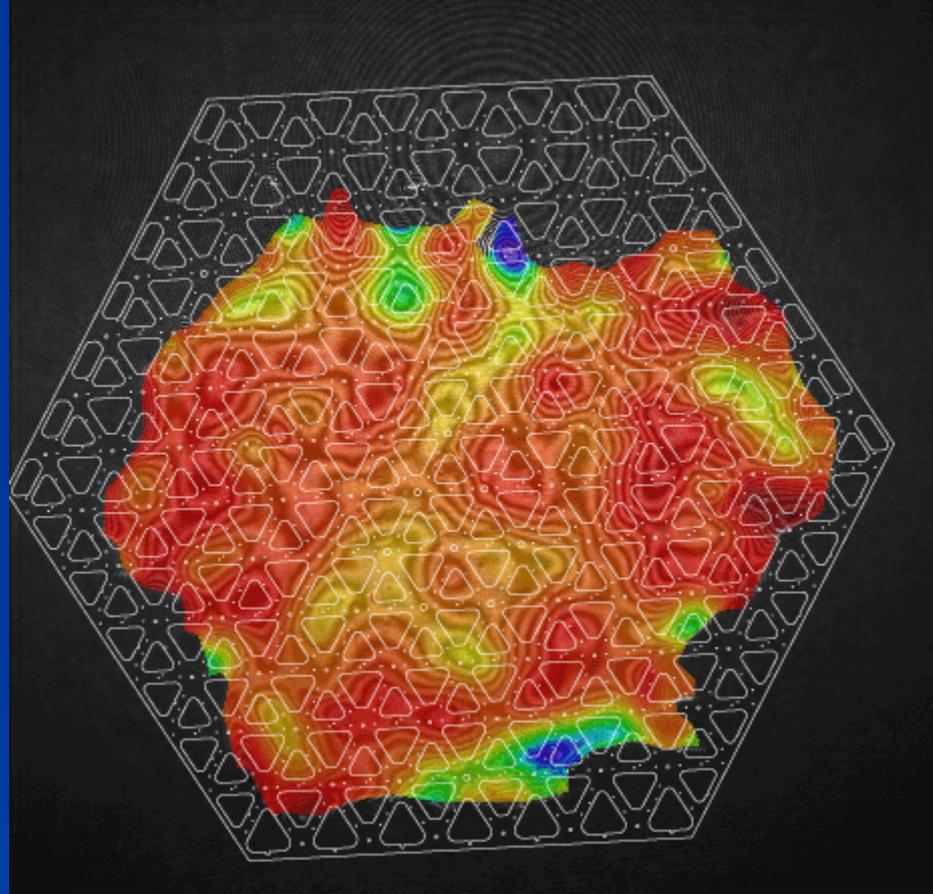


# Visible Interferometry

Most recent visible measurement

9 wvs P-V  
1.1 wvs RMS  
(HeNe 633 nm)

5.7  $\mu\text{m}$  P-V  
0.7  $\mu\text{m}$  RMS





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# Current Status

One final push to improve the figure

Characterize the errors in the system

Measure actuator behavior

Measure mirror stability



# Lessons Learned

Scaling a half-meter design up to a two-meter model is not as easy as it sounds on paper!

Handling becomes a major engineering challenge

More than one measurement scheme is required

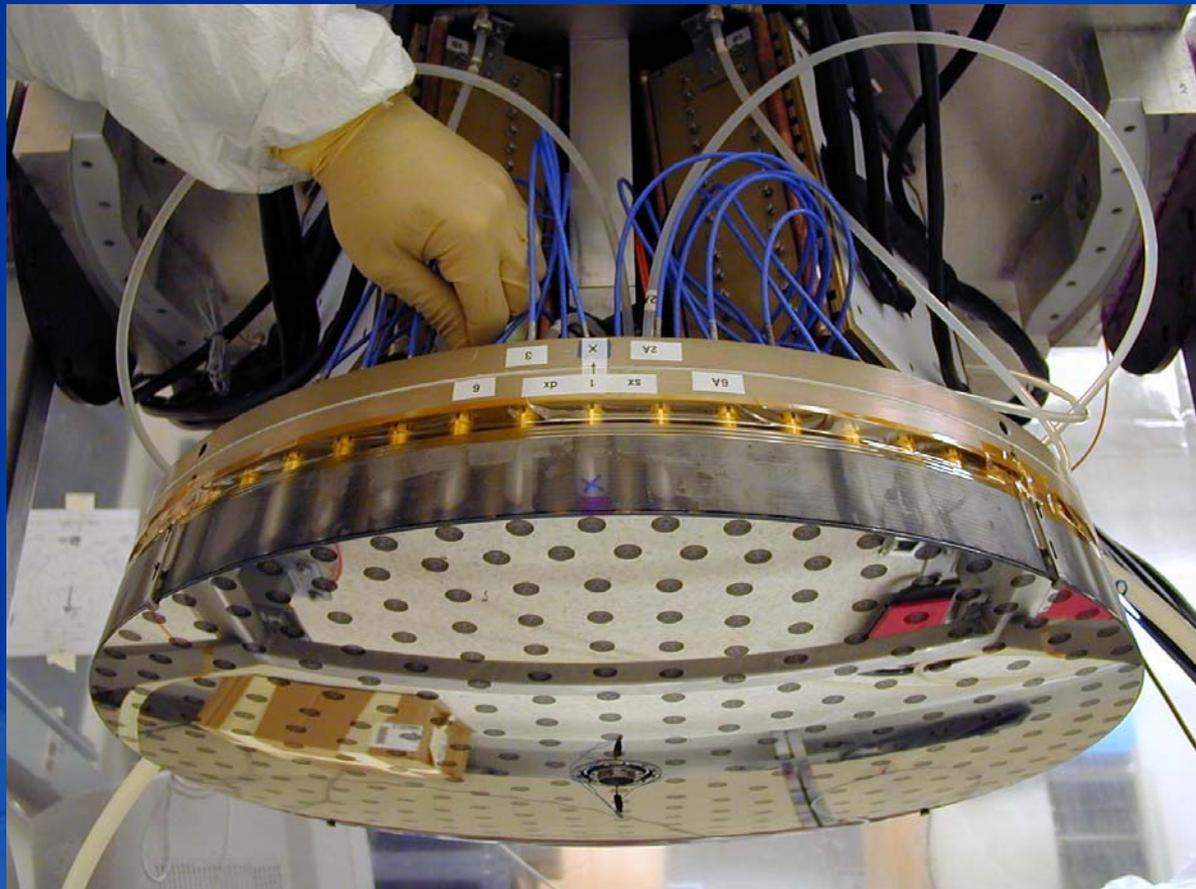
Distance encoders on the actuators would save a lot of time and money

Actuators with `run' and `walk' modes would be helpful

4D Technology's PhaseCam was crucial

# Technology Benefits

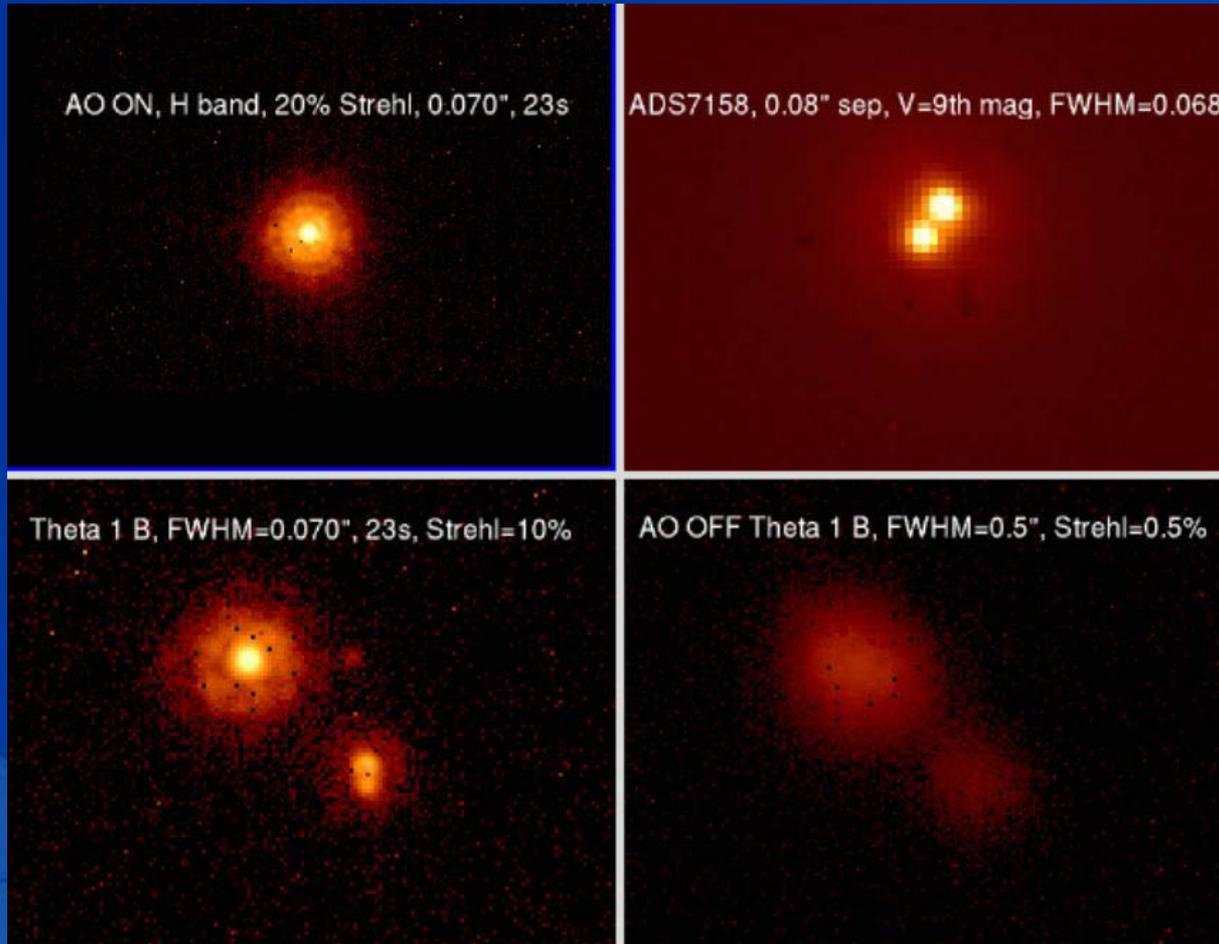
Immediate future: Experience gained with fabricating thin optics has led to successful AO secondary mirrors



Multiple Mirror Telescope (MMT) Adaptive Secondary Mirror

# Technology Benefits

MMT results with and without adaptive secondary



# Current Arizona Research



membrane  
thickness

number of  
actuators

What's the most efficient way to distribute the mass  
The minimum surface error occurs when...  
for the best quality surface?

**Membrane mass = Total actuator mass**

# In Conclusion

Visible interferometry results are in.

Fabrication experience has already been put to good use.

Experience gained with 2-m, 15kg/m<sup>2</sup> class mirrors will be helpful for post-JWST imaging systems.

Currently working on a design algorithm that produces the most accurate mirror for the lightest mass.



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